

Problem 4A.5

Numerical demonstration of the von Kármán momentum balance.

- (a) Evaluate the integrals in Eq. 4.4-13 numerically for the Blasius velocity profile given in Fig. 4.4-3.
- (b) Use the results of (a) to determine the magnitude of the wall shear stress $\tau_{yx}|_{y=0}$.
- (c) Calculate the total drag force, F_x , for a plate of width W and length L , wetted on both sides. Compare your result with that obtained in Eq. 4.4-30.

Answers: (a) $\int_0^\infty \rho v_x (v_e - v_x) dy = 0.664 \sqrt{\rho \mu v_\infty^3 x}$
 $\int_0^\infty \rho (v_e - v_x) dy = 1.73 \sqrt{\rho \mu v_\infty x}$

Solution

Part (a)

Eq. 4.4-13 on page 136 is the fabled von Kármán momentum balance.

$$\mu \left. \frac{\partial v_x}{\partial y} \right|_{y=0} = \frac{d}{dx} \int_0^\infty \rho v_x (v_e - v_x) dy + \frac{dv_e}{dx} \int_0^\infty \rho (v_e - v_x) dy \quad (4.4-13)$$

Note that v_x represents the velocity distribution inside the boundary layer, and v_e represents the velocity external to the boundary layer. For tangential laminar flow along a flat plate in particular, $v_e = v_\infty$ and $v_x/v_\infty = f'$, where

$$f = f(\eta) = f \left(y \sqrt{\frac{v_\infty}{\nu x}} \right)$$

satisfies the Blasius equation,

$$-f f'' = 2f''',$$

subject to the following boundary conditions.

$$\begin{aligned} f(0) &= 0 \\ f'(0) &= 0 \\ \lim_{\eta \rightarrow \infty} f'(\eta) &= 1 \end{aligned}$$

The two integrals in the von Kármán momentum balance can be written as

$$\begin{aligned} \int_0^\infty \rho v_x (v_\infty - v_x) dy & \quad \int_0^\infty \rho (v_\infty - v_x) dy \\ \int_0^\infty \rho v_\infty v_x \left(1 - \frac{v_x}{v_\infty} \right) dy & \quad \int_0^\infty \rho v_\infty \left(1 - \frac{v_x}{v_\infty} \right) dy, \end{aligned}$$

and the differential dy can be written in terms of $d\eta$ by

$$d\eta = dy \sqrt{\frac{v_\infty}{\nu x}} \quad \rightarrow \quad \sqrt{v_\infty} dy = \sqrt{\nu x} d\eta = \sqrt{\frac{\mu x}{\rho}} d\eta.$$

As a result,

$$\begin{aligned}
 \int_0^\infty \rho \sqrt{v_\infty} v_x \left(1 - \frac{v_x}{v_\infty}\right) \sqrt{v_\infty} dy &= \int_0^\infty \rho \sqrt{v_\infty} \left(1 - \frac{v_x}{v_\infty}\right) \sqrt{v_\infty} dy, \\
 \int_0^\infty \rho \sqrt{v_\infty} v_x \left(1 - \frac{v_x}{v_\infty}\right) \sqrt{\frac{\mu x}{\rho}} d\eta &= \int_0^\infty \rho \sqrt{v_\infty} \left(1 - \frac{v_x}{v_\infty}\right) \sqrt{\frac{\mu x}{\rho}} d\eta \\
 \sqrt{\rho \mu v_\infty^3 x} \int_0^\infty v_x \left(1 - \frac{v_x}{v_\infty}\right) d\eta &= \sqrt{\rho \mu v_\infty^3 x} \int_0^\infty \left(1 - \frac{v_x}{v_\infty}\right) d\eta \\
 \sqrt{\rho \mu v_\infty^3 x} \int_0^\infty \frac{v_x}{v_\infty} \left(1 - \frac{v_x}{v_\infty}\right) d\eta &= \sqrt{\rho \mu v_\infty^3 x} \int_0^\infty (1 - f') d\eta \\
 \sqrt{\rho \mu v_\infty^3 x} \int_0^\infty f'(1 - f') d\eta &= \sqrt{\rho \mu v_\infty^3 x} \int_0^\infty (1 - f') d\eta.
 \end{aligned}$$

A plot of the numerical solution to the Blasius equation is shown in Fig. 4.4-3.

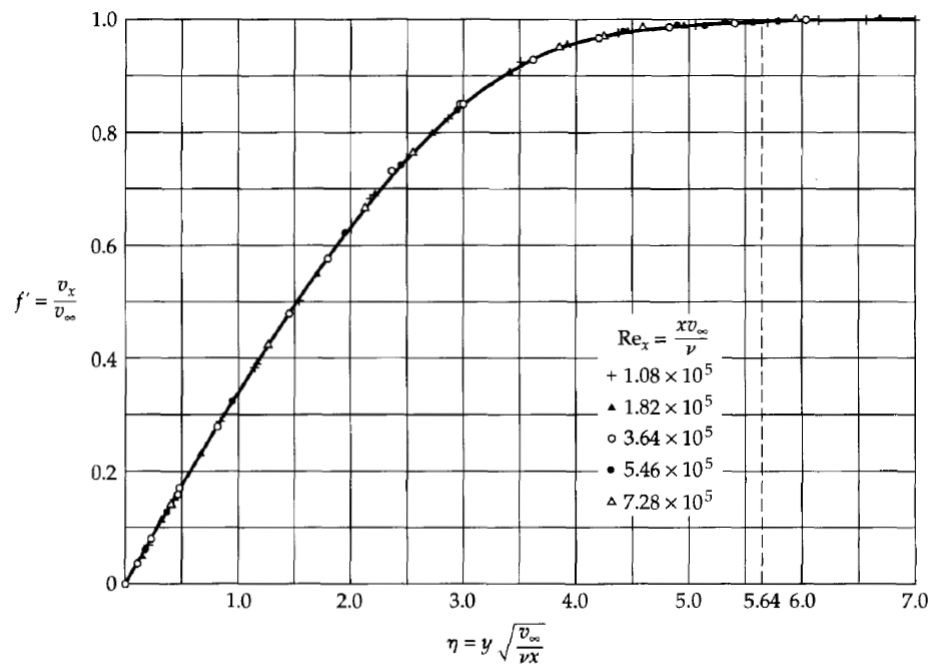


Fig. 4.4-3. Predicted and observed velocity profiles for tangential laminar flow along a flat plate. The solid line represents the solution of Eqs. 4.4-20 to 24, obtained by Blasius [see H. Schlichting, *Boundary-Layer Theory*, McGraw-Hill, New York, 7th edition (1979), p. 137].

According to this graph, $f' \approx 1$ after $\eta = 7.0$, so $1 - f' \approx 0$ after $\eta = 7.0$. The two integrals can then be approximated by

$$\sqrt{\rho \mu v_\infty^3 x} \int_0^{7.0} f'(1 - f') d\eta \quad \sqrt{\rho \mu v_\infty^3 x} \int_0^{7.0} (1 - f') d\eta.$$

To evaluate these integrals numerically, apply Simpson's rule, which states that

$$\int_a^b f(x) dx \approx \frac{\Delta x}{3} [f(x_0) + 4f(x_1) + 2f(x_2) + 4f(x_3) + \cdots + 2f(x_{n-2}) + 4f(x_{n-1}) + f(x_n)],$$

where $\Delta x = (b - a)/n$ and n is even.

Therefore, taking $\Delta\eta$ to be 0.5,

$$\begin{aligned} \int_0^\infty \rho v_x (v_e - v_x) dy &\approx \sqrt{\rho\mu v_\infty^3 x} \int_0^{7.0} f'(1-f') d\eta \\ &\approx \sqrt{\rho\mu v_\infty^3 x} \cdot \frac{7.0-0}{3} [0(1-0) + 4(0.17)(1-0.17) + 2(0.34)(1-0.34) + 4(0.49)(1-0.49) \\ &\quad + 2(0.63)(1-0.63) + 4(0.75)(1-0.75) + 2(0.85)(1-0.85) \\ &\quad + 4(0.92)(1-0.92) + 2(0.96)(1-0.96) + 4(0.98)(1-0.98) \\ &\quad + 2(0.99)(1-0.99) + 4(0.999)(1-0.999) + 2(0.9999)(1-0.9999) \\ &\quad + 4(0.99999)(1-0.99999) + 0.999999(1-0.999999)] \\ &\approx 0.6596\sqrt{\rho\mu v_\infty^3 x} \end{aligned}$$

and

$$\begin{aligned} \int_0^\infty \rho(v_e - v_x) dy &\approx \sqrt{\rho\mu v_\infty x} \int_0^{7.0} (1-f') d\eta \\ &\approx \sqrt{\rho\mu v_\infty x} \cdot \frac{7.0-0}{3} [(1-0) + 4(1-0.17) + 2(1-0.34) + 4(1-0.49) \\ &\quad + 2(1-0.63) + 4(1-0.75) + 2(1-0.85) \\ &\quad + 4(1-0.92) + 2(1-0.96) + 4(1-0.98) \\ &\quad + 2(1-0.99) + 4(1-0.999) + 2(1-0.9999) \\ &\quad + 4(1-0.99999) + (1-0.999999)] \\ &\approx 1.704\sqrt{\rho\mu v_\infty x}. \end{aligned}$$

These results are accurate to two significant figures. For better accuracy, use a table for the values of f' rather than trying to guess them from the graph in Fig. 4.4-3. Consult page 139 of the seventh edition of Schlichting's book, "Boundary Layer Theory," for example.

Part (b)

From Appendix B.1 on page 843, the shear stress τ_{yx} is

$$\tau_{yx} = -\mu \left[\frac{\partial v_y}{\partial x} + \frac{\partial v_x}{\partial y} \right],$$

but because the flow along a flat plate only moves horizontally in the x -direction, it reduces to

$$\tau_{yx} = -\mu \frac{\partial v_x}{\partial y}.$$

Take the absolute value of both sides and substitute this result into the left side of the von Kármán momentum balance.

$$\begin{aligned} |\tau_{yx}|_{y=0} &= \frac{d}{dx} \int_0^\infty \rho v_x (v_e - v_x) dy + \frac{dv_e}{dx} \int_0^\infty \rho (v_e - v_x) dy \\ &\approx \frac{d}{dx} (0.66\sqrt{\rho\mu v_\infty^3 x}) + \frac{dv_\infty}{dx} (1.7\sqrt{\rho\mu v_\infty x}) \\ &= 0.33\sqrt{\frac{\rho\mu v_\infty^3}{x}} \end{aligned}$$

Part (c)

To obtain the total drag force, integrate the shear stress over the area of the flat plate. A factor of 2 is included since the plate is wetted on both sides.

$$\begin{aligned} F_x &= 2 \int |\tau_{yx}|_{y=0} dA \\ &= 2 \int_0^L |\tau_{yx}|_{y=0} (W dx) \\ &= 2 \int_0^L 0.33 \sqrt{\frac{\rho \mu v_\infty^3}{x}} (W dx) \\ &= 2 \times 0.33 \sqrt{\rho \mu v_\infty^3} W \int_0^L \frac{dx}{\sqrt{x}} \\ &= 2 \times 0.33 \sqrt{\rho \mu v_\infty^3} W (2\sqrt{L}) \\ &\approx 1.3 \sqrt{\rho \mu L W^2 v_\infty^3} \end{aligned}$$

This is very close to the result in Eq. 4.4-30, which has a coefficient of 1.328. The percent difference is

$$\frac{1.3 - 1.328}{1.328} \times 100\% \approx -2.1\%.$$